

A Gravity Balanced Test Stand For Flight Testing of MAVs

IMECE2008-78987

QI LU

Department of Mechanical and Aerospace
Engineering,
New Mexico State University,
Las Cruces, NM 88001,
qilu@nmsu.edu

CARLOS E. ORTEGA

Department of Mechanical and Aerospace
Engineering,
New Mexico State University,
Las Cruces, NM 88001,
ferrum00@hotmail.com

OU MA

Department of Mechanical and Aerospace Engineering
New Mexico State University,
Las Cruces, NM 88001,
oma@nmsu.edu

Overview

- Purpose of the test stand
- Innovations in our test stand
- Gravity Balancing Theory
- Simulation of the test stand

Purpose of The Testing Stand

- No Crashing, Safety, Simple Operation,



Simulation Testing



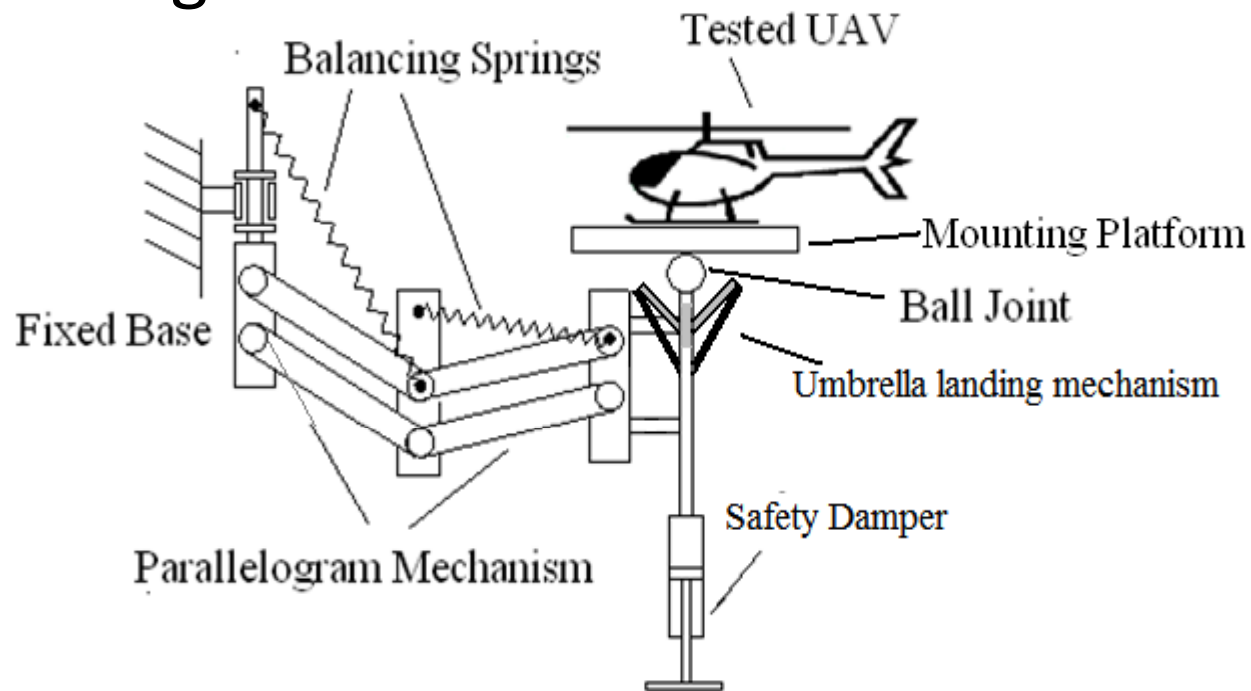
Testing with a Stand



Outdoor Testing

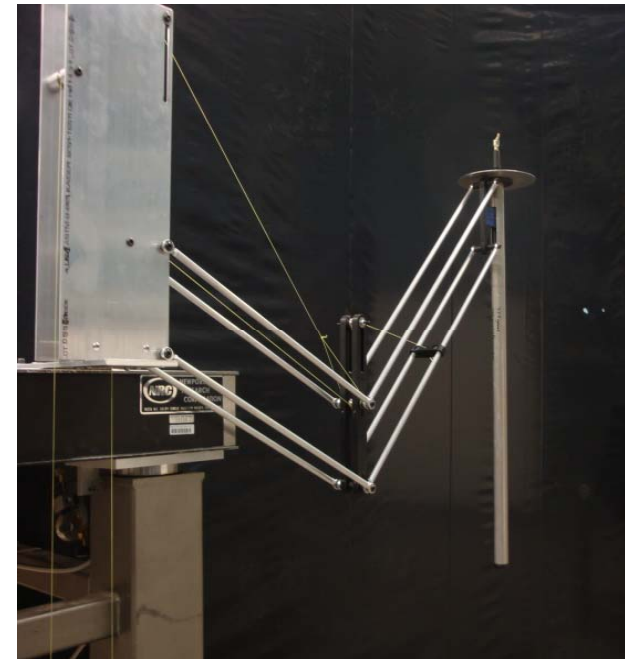
Innovations in Our Test Stand

- Higher Degree of freedom
- Up Righting Mechanism
- Gravity Balancing

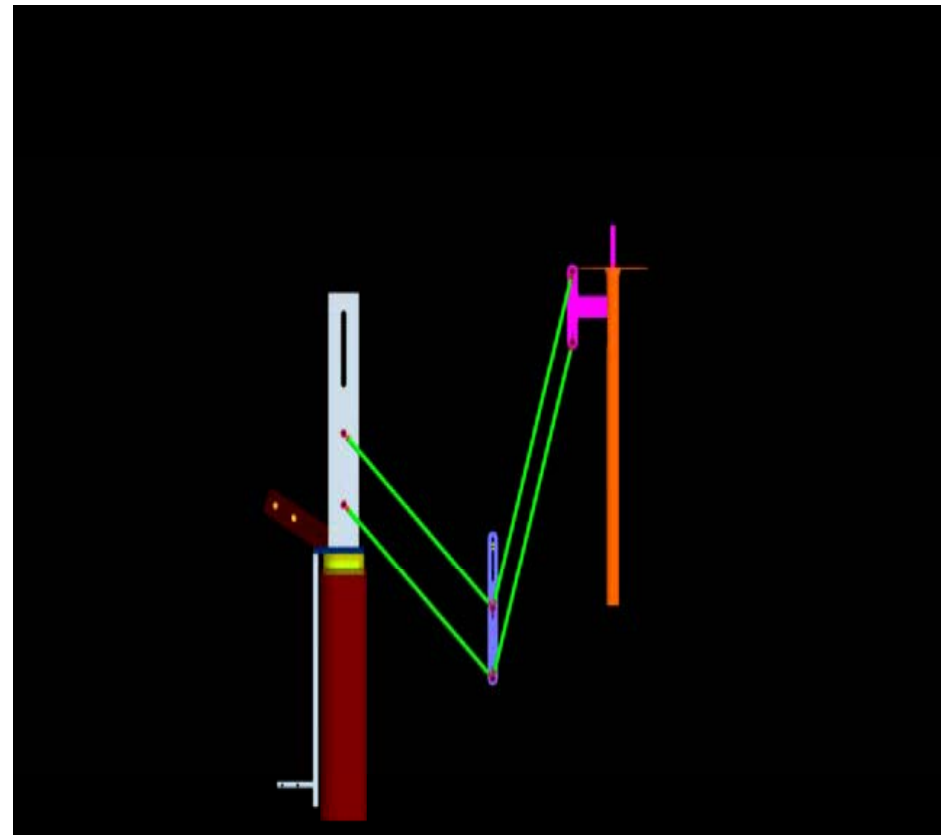
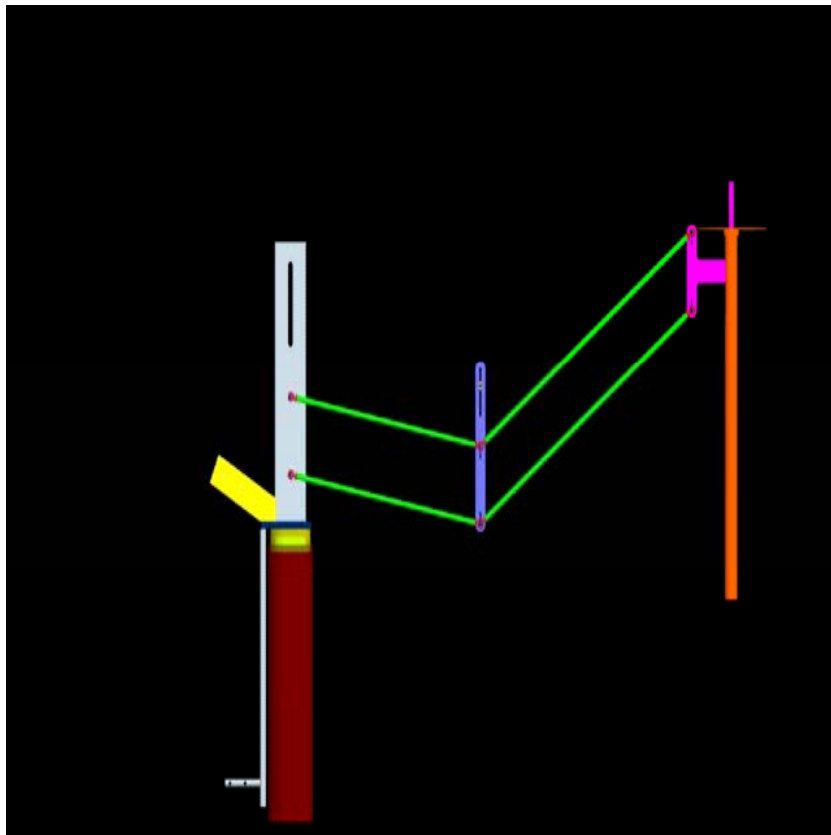


Innovations in Our Test Stand

- Higher degree of freedom

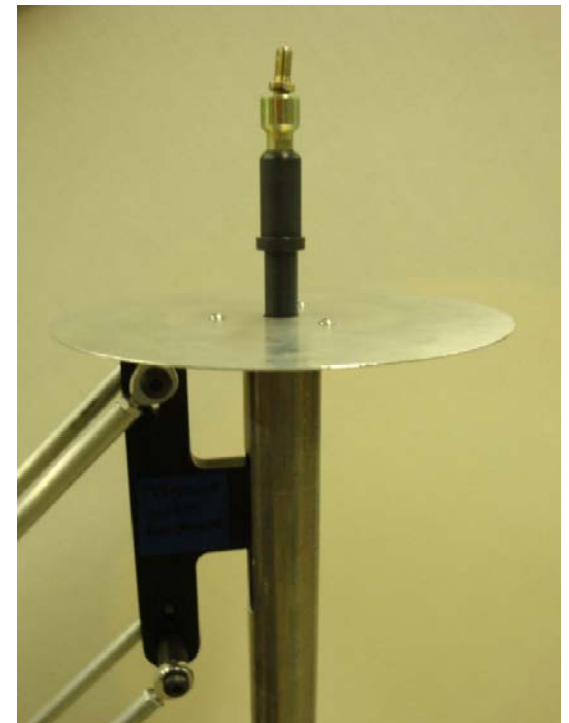
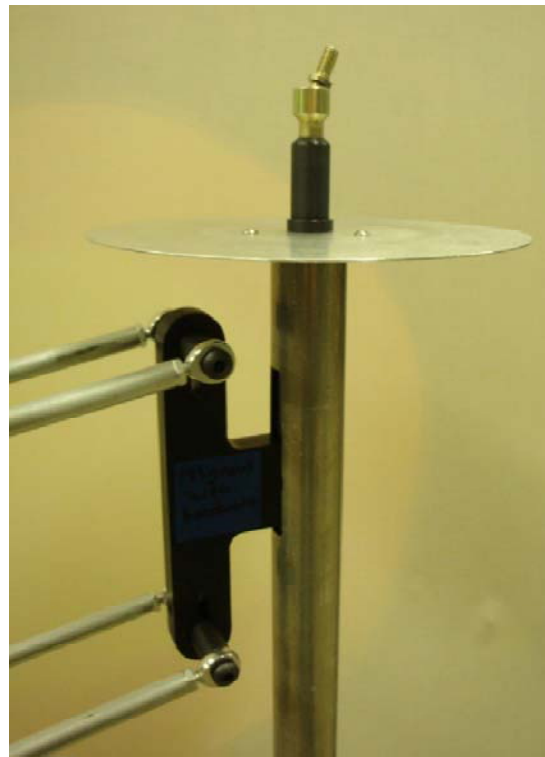
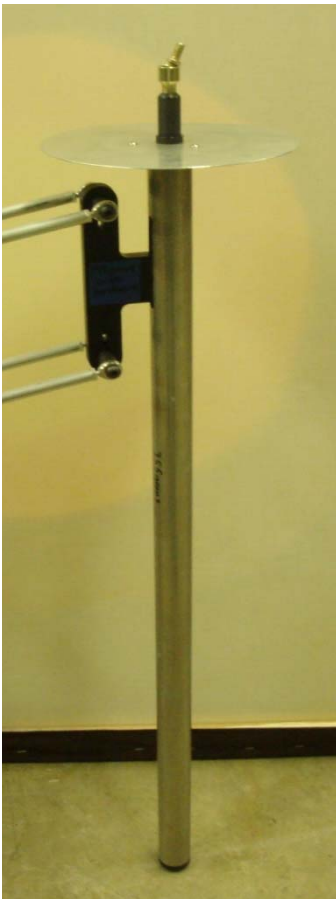


Innovations in Our Test Stand



Innovations in Our Test Stand

- Up righting mechanism



Innovations in Our Test Stand

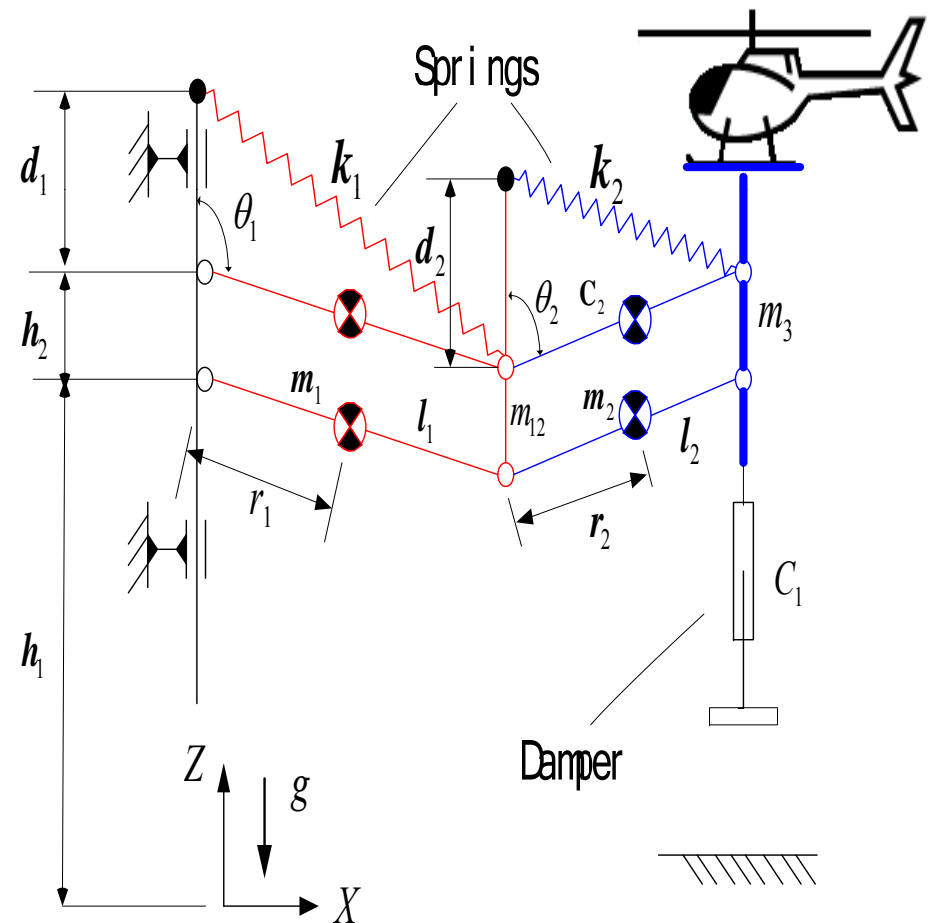
- Gravity Balancing

$$V_{p1gm1} = 2(h_1 + r_1 \cos \theta_1)m_1g + h_2m_1g + (h_1 + l_1 \cos \theta_1 + r_{12})m_{12}g$$

$$V_{p1gm2} = 2(h_1 + l_1 \cos \theta_1 + r_2 \cos \theta_2)m_2g + h_2m_2g + (h_1 + l_1 \cos \theta_1 + l_2 \cos \theta_2 + r_3)m_3g$$

$$V_{s1} = \frac{1}{2}k_1(d_1^2 + l_1^2 - 2d_1l_1 \cos \theta_1)$$

$$V_{s2} = \frac{1}{2}k_2(d_2^2 + l_2^2 - 2d_2l_2 \cos \theta_2)$$

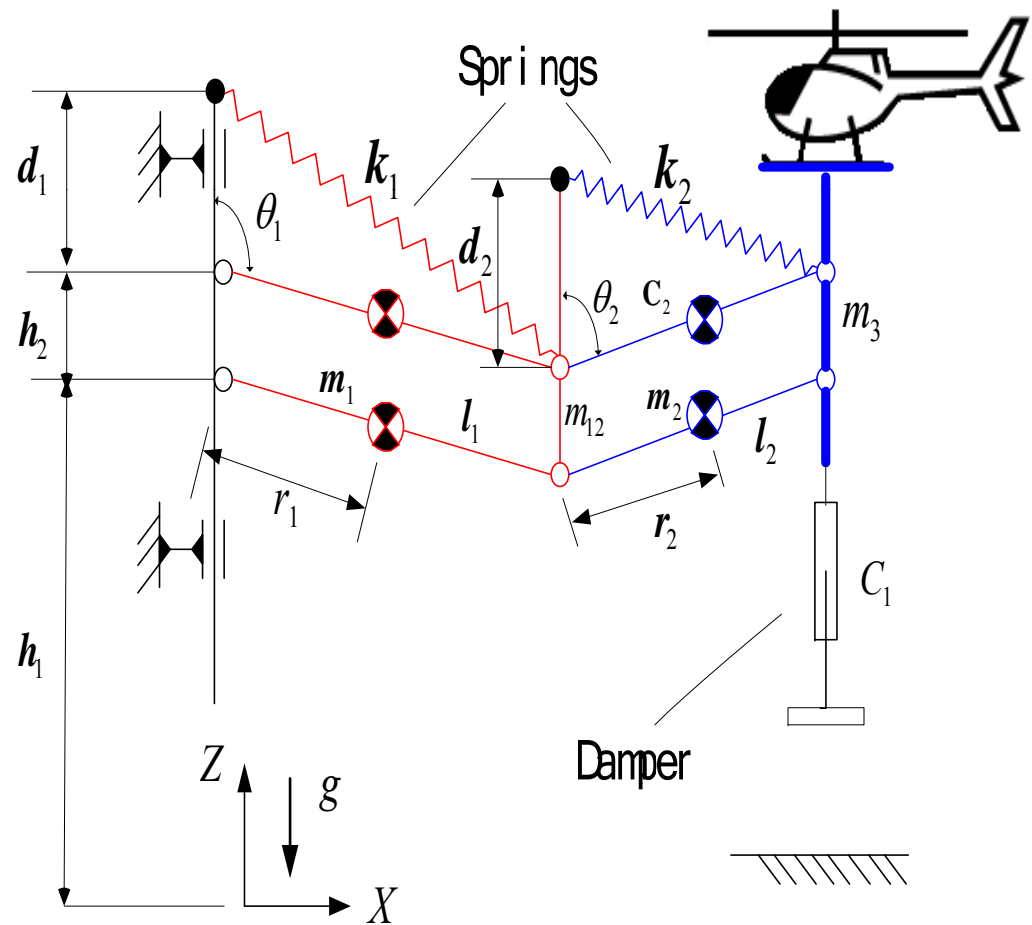


Gravity Balancing Theory

- Spring constants

$$k_1 = \frac{2m_1 r_1 + l_1(2m_2 + m_{12} + m_3)}{l_1 d_1} g$$

$$k_2 = \frac{2m_2 r_2 + l_2 m_3}{l_2 d_2} g$$

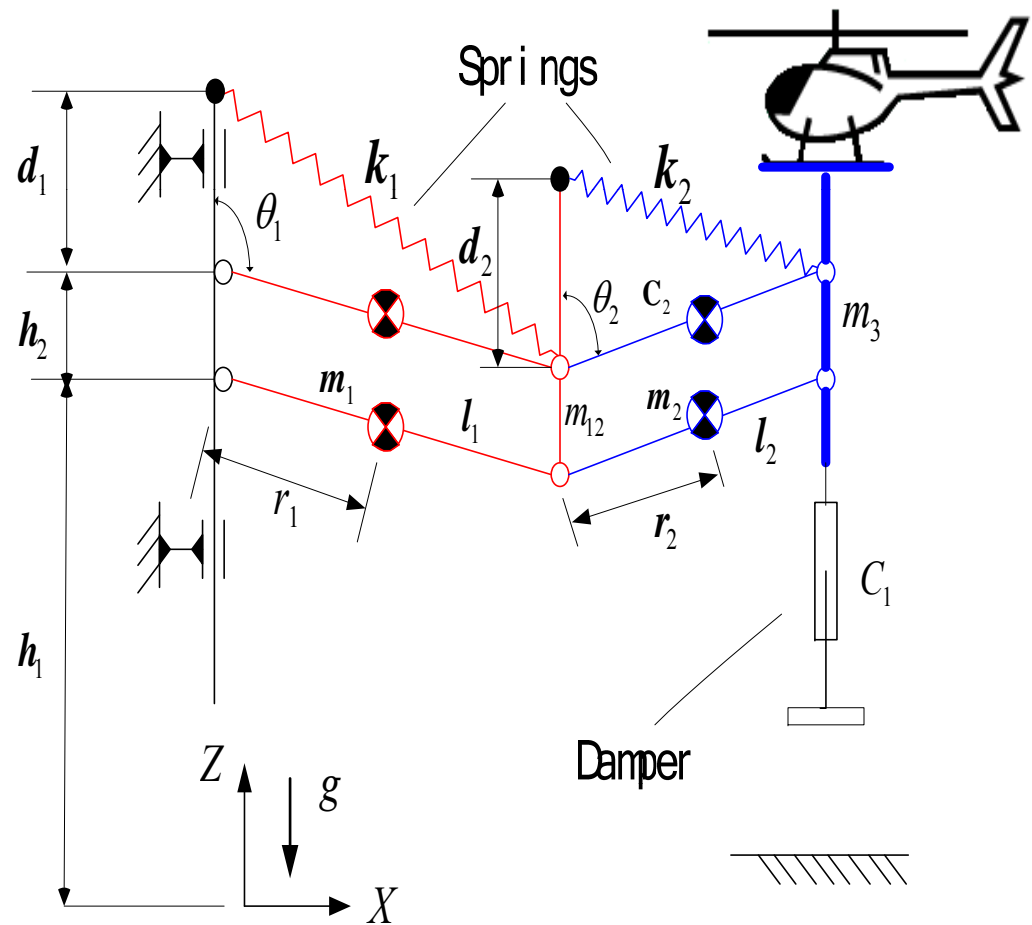


Gravity Balancing Theory

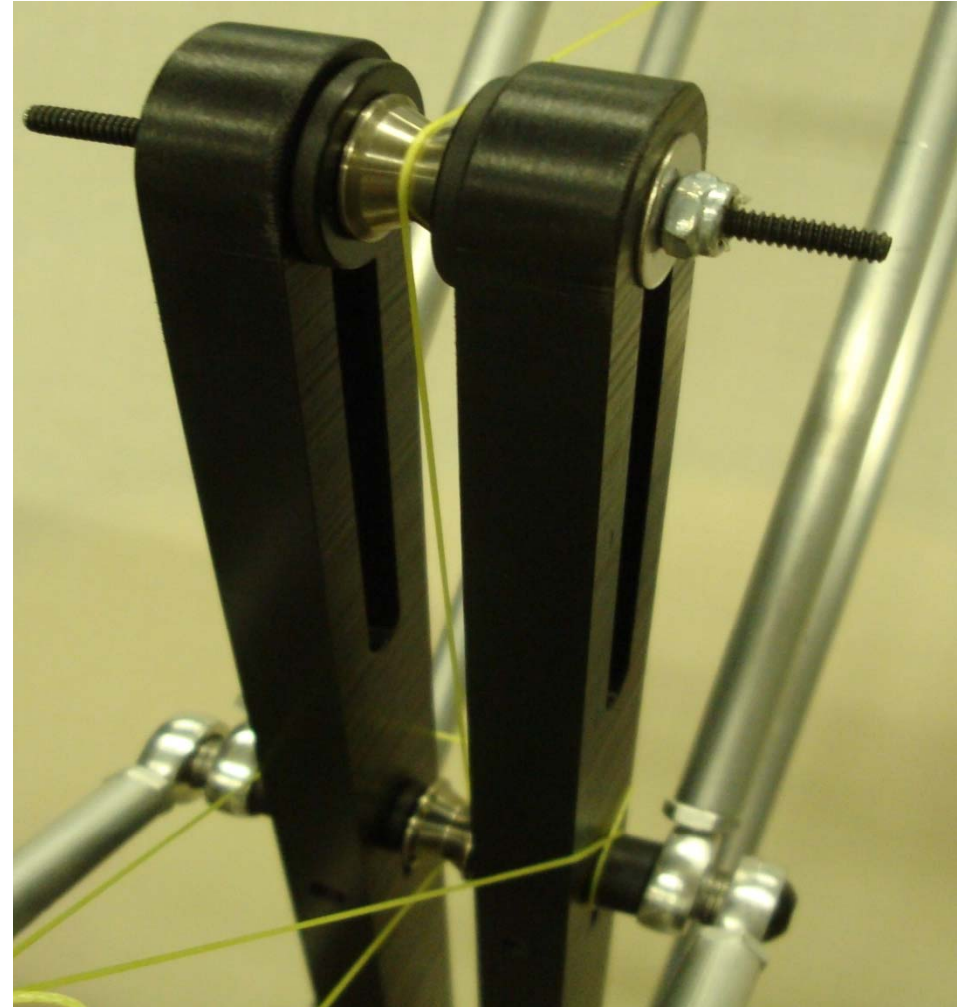
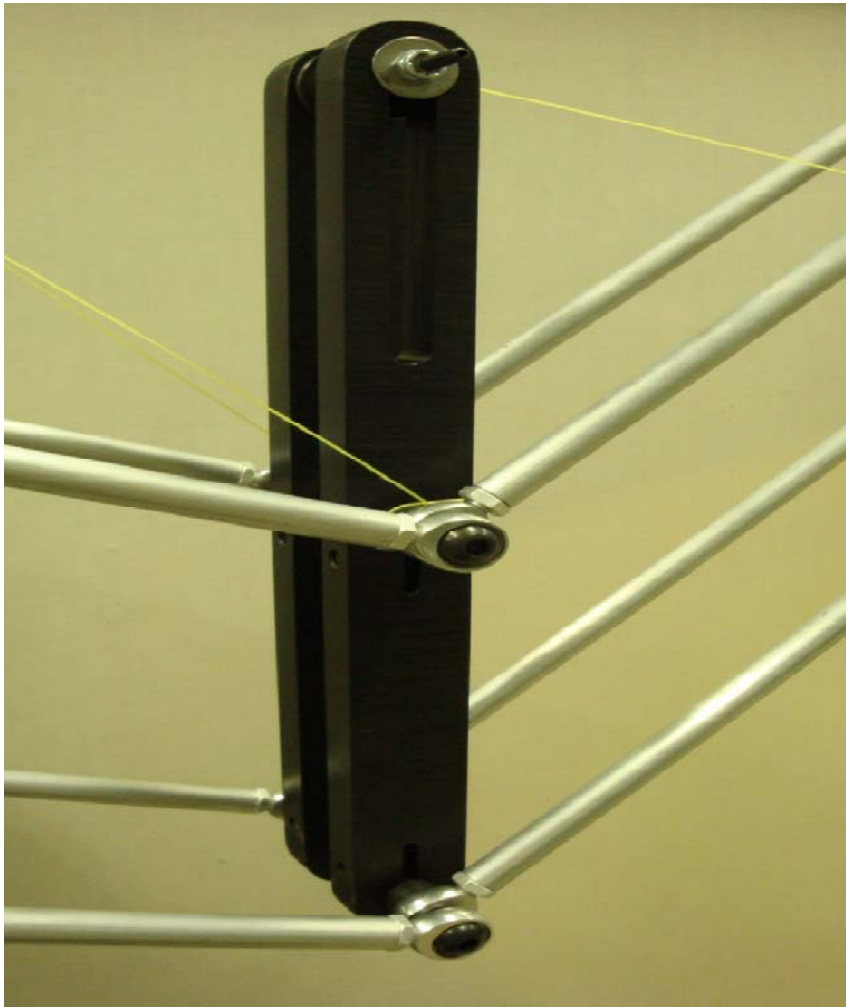
- Adjustability

$$d_1 = \frac{2m_1 r_1 + l_1(2m_2 + m_{12} + m_3)}{l_1 k_1} g$$

$$d_2 = \frac{2m_2 r_2 + l_2 m_3}{l_2 k_2} g$$

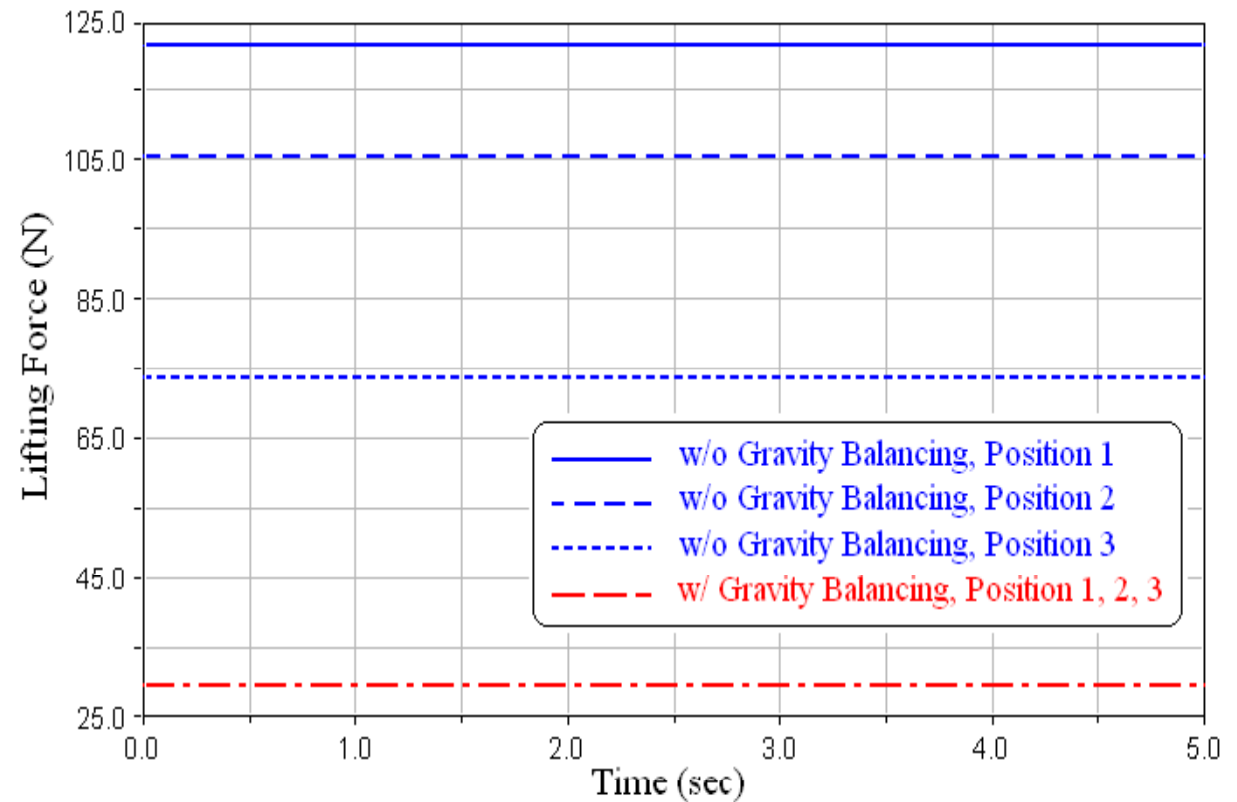
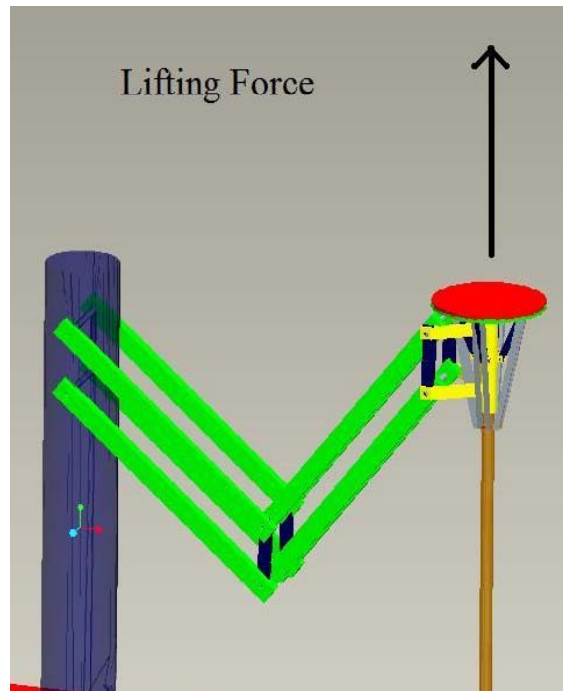


Gravity Balancing Theory



Simulation

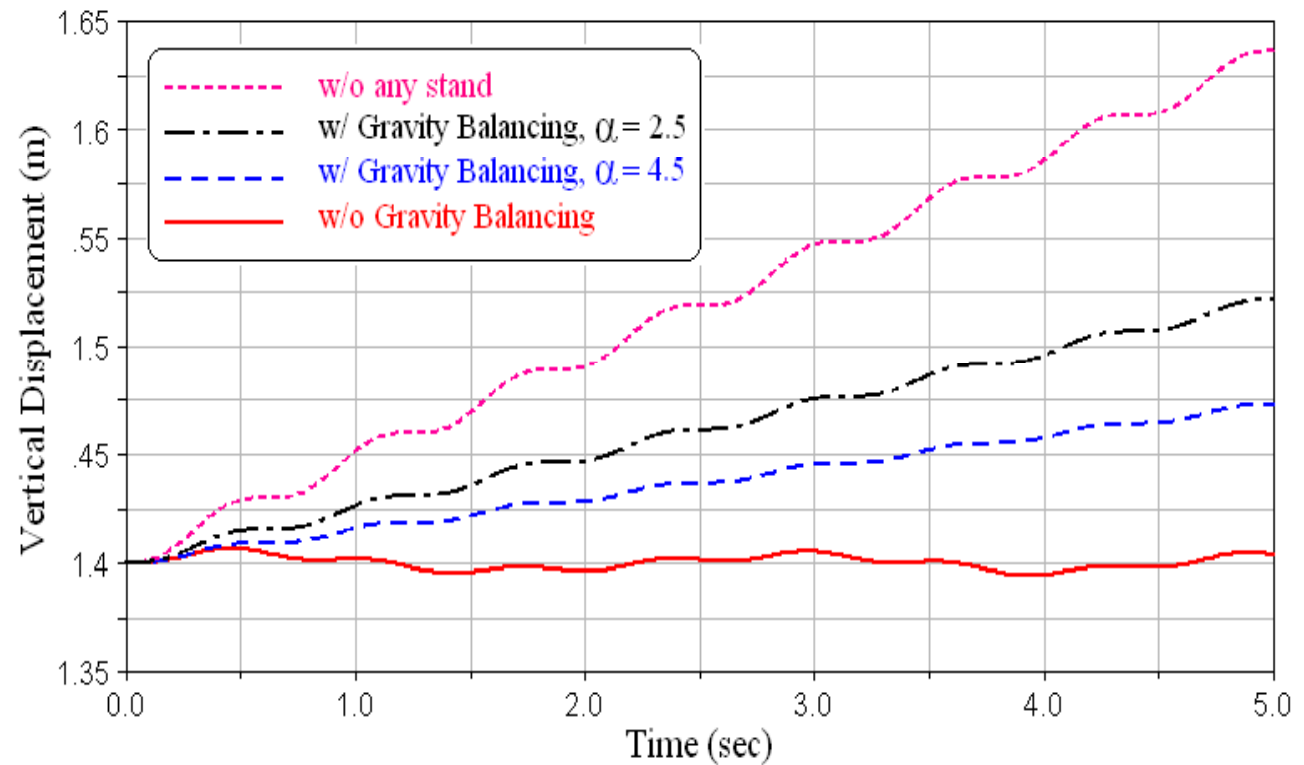
- Required lifting force



Simulation

- Vertical Displacement

$$\alpha = \frac{\text{mass of test stand}}{\text{mass of MAV}}$$



Conclusions and Further Work

- Gravity balancing is necessary for MAVs.
- Design has been done.
- Hardware prototyping is underway.
 - Complete the hardware prototype.
 - Verify the gravity balancing capability.
 - Investigate dynamic behavior.

Questions?

Many Thanks to

- Physical Science Lab / Air Force (Sponsor)
- NMSU Aerospace Program
- NASA Space Grant